

Aircraft Spar Beam Torsion

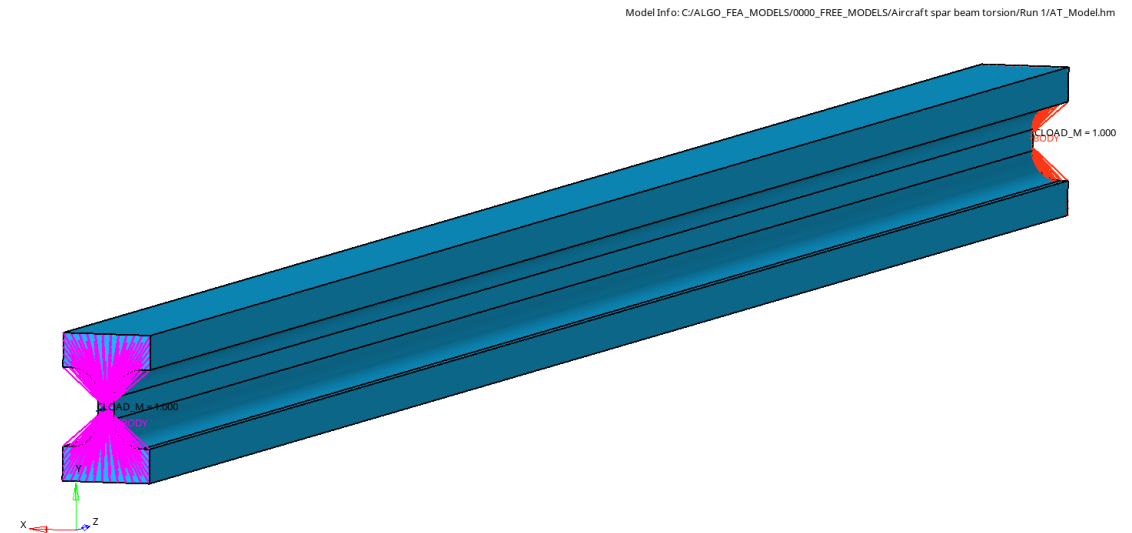
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For questions, please fill out contact form



Model Description

- Torsion analysis of a spar beam based on example question 2 documented in page 385 of the following book :
 - J. Souza, *Roark's Formulas For Stress And Strain-.pdf*. Accessed: Nov. 06, 2022. [Online]. Available: https://www.academia.edu/37205286/Roarks_Formulas_For_Stress_And_Strain_pdf
- As mentioned in the book, spar beam is made out of spruce wood.



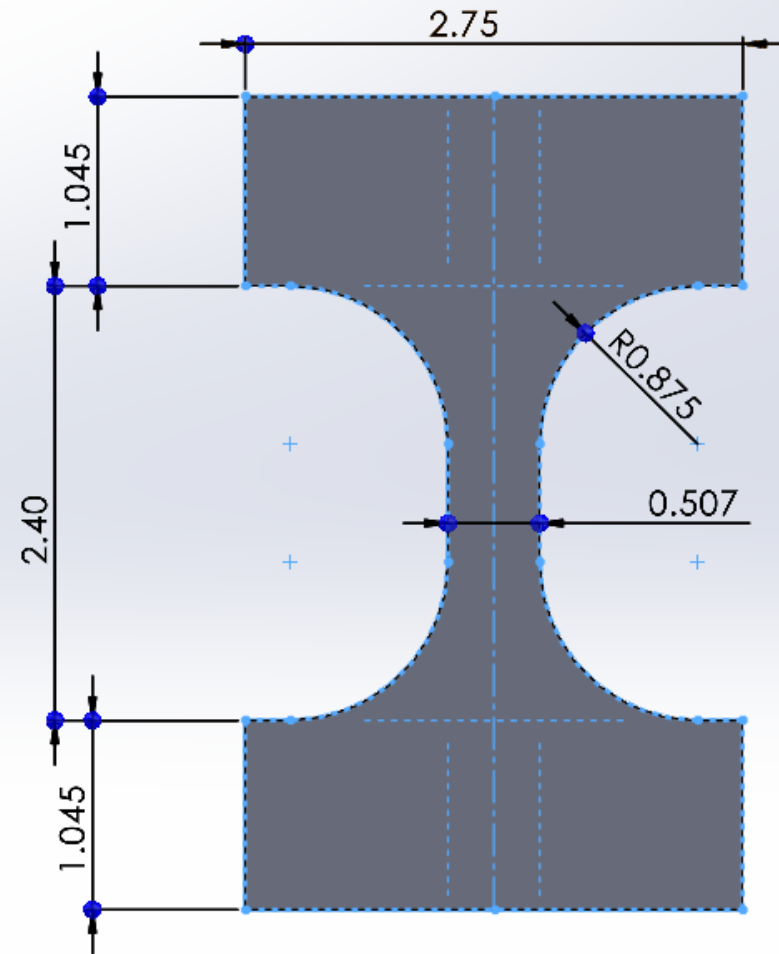
Model Description

Length – 8 ft

Young's modulus – 1500000 lb/in²

Modulus of rigidity – 100000 lb/in²

Poisson's ratio – 0.38



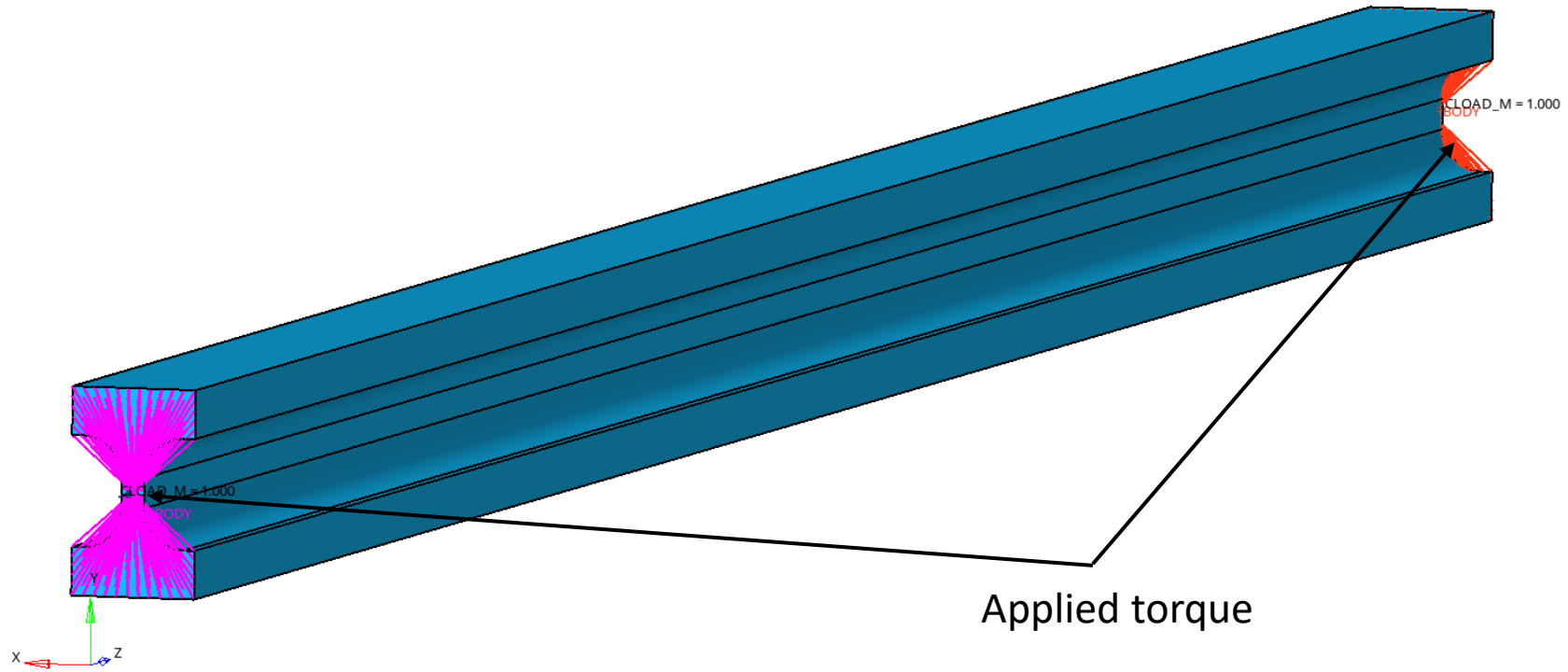
Model Parameters

Entity	Type
Solver	Altair Radioss
Version	2021.2.1
Processors	2
Threads	2
CPU	Intel(R) Core(TM) i7-9750H CPU @ 2.60GHz
Total run time	1 sec

FEA Entities	Type
Analysis Type	Dynamic Explicit
Unit System	lb, in, s
Element Type	HEXA8N
Material Type	M1_ELAST
Property Type	P14_SOLID

Analysis Setup

Model Info: C:/ALGO_FEA_MODELS/0000_FREE_MODELS/Aircraft spar beam torsion/Run 1/AT_Model.hm



Analysis Assumptions and Limitations

- Standard density value and standard Poisson's ratio of spruce is assumed.

Hand Calculations

K – Length to the fourth function of cross section E
 – Young's modulus ν – Poisson's ratio T – torque l
 – length G – modulus of Rigidity A – Area of the section

$$K = 2K_1 + K_2 + 2\alpha D^4$$

$$K_1 = ab^3 \left[\frac{1}{3} - 0.21 \frac{b}{a} \left(1 - \frac{b^4}{12a^4} \right) \right] = 0.796 \text{ in}^4$$

$$K_2 = \frac{1}{3} cd^3 = 0.104 \text{ in}^4$$

$$\alpha = \frac{t^3}{t_1} \left(0.15 + 0.1 \frac{r}{b} \right) = 0.1133$$

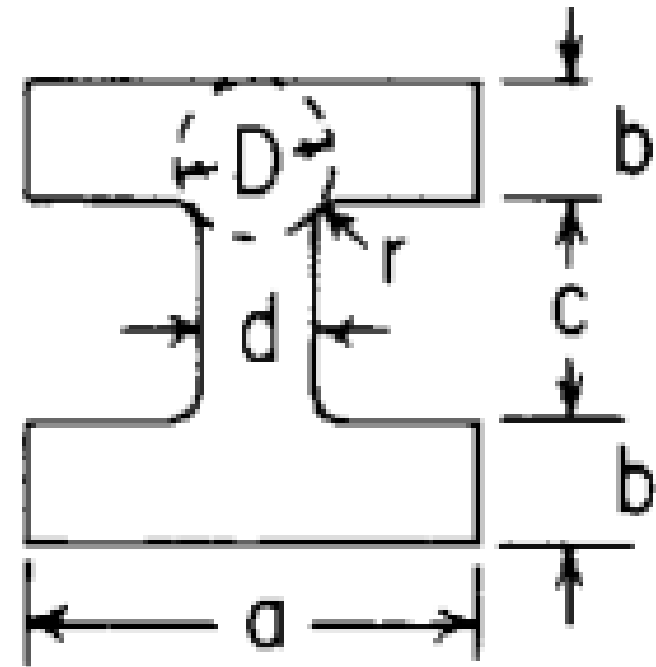


Figure – from above book

Hand Calculations

Here, $t_1 = d$

$$\theta = \frac{Tl}{KG} = 0.168 \text{ rad}$$

$$\tau_{max} = \frac{T}{K} C$$

Where,

$$C = \frac{D}{1 + \frac{\pi^2 D^4}{16A^2}} \left[1 + 0.15 \left(\frac{\pi^2 D^4}{16A^2} - \frac{D}{2r} \right) \right] = 1.73 \text{ in}$$

Therefore,

$$\tau_{max} = 303 \text{ lb/in}^2$$

Analysis Results

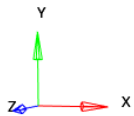
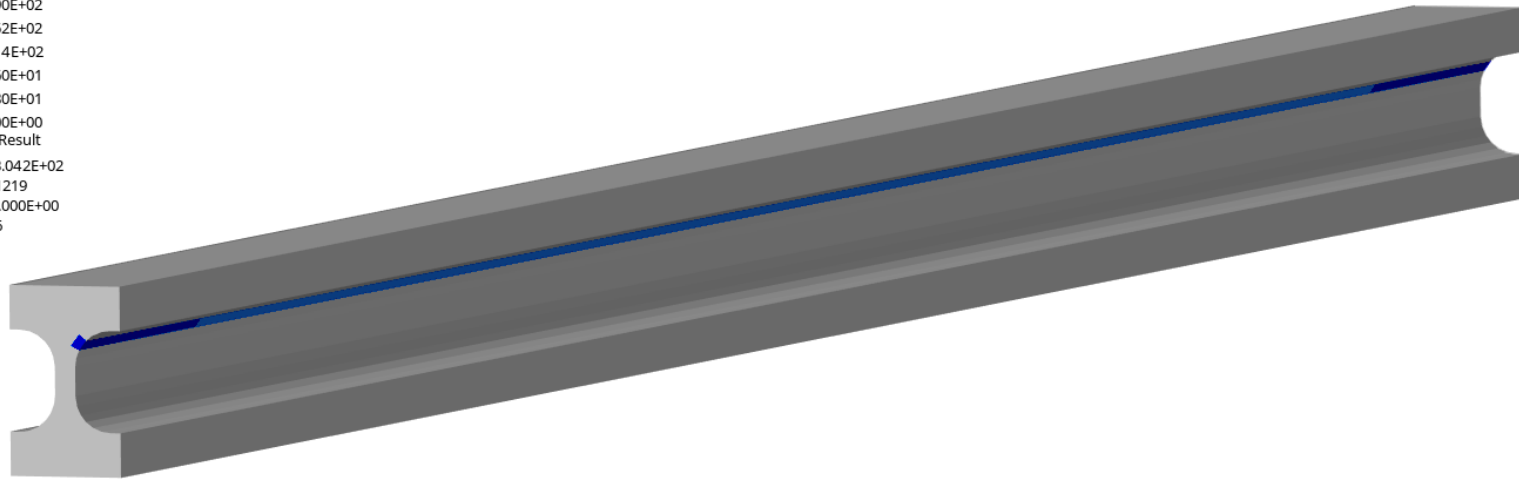
Units – lb/in²

Contour Plot
Von Mises(Scalar value)
Simple Average

3.042E+02
2.704E+02
2.366E+02
2.028E+02
1.690E+02
1.352E+02
1.014E+02
6.760E+01
3.380E+01
0.000E+00
No Result

Max = 3.042E+02
Node 11219
Min = 0.000E+00
Node 36

1: AT_Model
Loadcase 1 : Time = 1.3000e-01 : Frame 14



Analysis Results

- Maximum stress,
Based on hand calculations – 303 lb/in²
From the simulation – 304.2 lb/in²

Conclusions

- Torsion analysis of a spar beam conducted using Altair Radioss based on the book listed in slide 2.
- Results of the simulation correlate well to the expected hand calculation value.